TOWARDS IMPROVED SPRAY MODELS FOR HANDLING COMPLEX ENGINE GEOMETRIES

Federico Perinia, Rolf D. Reitza

^aUniversity of Wisconsin-Madison

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Summary - Motivation

- Internal injector scales can differ from combustion engine scales by 2-3 orders of magnitude
- Realistic engine geometries require non-axial grids and the limited flow field resolution near the injector can affect spray simulations

Reduce the dependency of spray predictions on grid resolution through an improved sub-grid-scale representation of the spray jet flow field

Improve the computational efficiency of the methodology to adopt more realistic numbers of particles

Long-term goal: Predictive multi-physics modeling of ICE flows, sprays, and combustion



Efficient grid-independent collision modeling

- Tetrahedralization of the drop-in-parcel representation for fast ROI calculation
 - Deterministic collision parameter estimation allows efficient pre-processing using data-mining techniques



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Sub-grid scale modeling of transient gas-jet

A time-resolved sub-grid scale gas-jet model to cope with underresolved grid near the nozzle

$$\begin{cases} \boldsymbol{\Theta}_{B} = \frac{\boldsymbol{\Theta}_{n}' + \Delta t \, d_{p,sgs} \left(\boldsymbol{\Theta}_{t} + \mathbf{u}_{sgs}\right)}{1 + \Delta t \, d_{p,sgs}} \\ \left(m_{B} + S_{uvw}\right) \mathbf{u}_{B} - m_{n} \mathbf{u}_{n} = \mathbf{E} - \mathbf{r}_{u} \\ \downarrow \end{cases}$$

$$\begin{split} S_{uvw} &= \sum_{p \in i4} \begin{cases} \frac{4}{3} \pi \rho_p N_p r_B^3 \frac{\Delta t \, d_p}{1 + \Delta t \, d_p}, & p \notin \Omega_{\text{jet}} \\ 0, & p \in \Omega_{\text{jet}} \end{cases} \\ \mathbf{r}_u &= \sum_{p \in i4} \begin{cases} \frac{4}{3} \pi \rho_p N_p \left(r_B^3 \frac{\mathbf{\theta}'_n + \Delta t \, d_p \, \mathbf{\theta}_t}{1 + \Delta t \, d_p} - r_n^3 \mathbf{\theta}'_n \right), & p \notin \Omega_{\text{jet}} \\ \frac{4}{3} \pi \rho_p N_p \left(r_B^3 \frac{\mathbf{\theta}'_n + \Delta t \, d_p \left(\mathbf{\theta}_t + \mathbf{u}_{sgs}\right)}{1 + \Delta t \, d_p} - r_n^3 \mathbf{\theta}'_n \right) & p \in \Omega_{\text{jet}} \end{cases} \end{split}$$



Solution within the jet region provided by the analogy $\frac{f_v x}{u} = Strouhal \approx Stokes = \frac{\tau_I}{\tau_p}$



slide 4

Sub-grid scale modeling of transient gas-jet

Transient turbulent gas-jet velocity field of Abani and Reitz, PoF2007

$$\mathbf{u}_{axis}(x,t) = \mathbf{u}_{inj}(t_0) + \int_{t_0}^t \left[1 - \exp\left(-\frac{\tau - t_0}{St\left(x - x_{inj}\right)} \left\| \mathbf{u}_{inj}(\tau) \right\| \right) \right] \frac{\partial \mathbf{u}_{inj}(\tau)}{\partial \tau} d\tau,$$
$$\mathbf{u}_{sgs}(x,r,t) = \frac{f_{entr}(x)\mathbf{u}_{axis}(x,t)}{\left(1 + \frac{12r^2}{x^2K_{entr}^2}\right)^2}$$

Model parameters

- **St** = Stokes number = spatial response time
- \mathbf{K}_{entr} = Entrainment constant = turbulent dispersion due to viscosity
- \mathbf{f}_{entr} = how much of the momentum actually transfers to the gas phase



GA study of spray model constants

- Multiple interacting spray models \rightarrow hard to isolate the effects of each model
- Hard to validate each of these isolated phenomena against experiments
- May be aided by future DNS simulations

A GA optimization to answer these questions:

- \rightarrow What **parameter regions** should we move in?
- → When we used to calibrate the spray constants, how much were we tweaking the gas-phase prediction too?

→ Is there an **optimal calibration set**, and, does this include "historically used" values or does it suggest new ones, which better fit the newest and highly confident Sandia experiments?



GA study of spray model constants

6 model variables				
Variable	name	std value	range	to
RT time constant	C _{RT}	1.0	0.05	50.0
RT wavelength cnst.	CART	0.1	0.01	10.0
KH decay timescale cnst.	B ₁	40.0	10.0	100.0
Gas-jet Stokes number	St	3.0	0.1	5.0
Gas-jet entrainment cnst.	K _{entr}	0.45	0.3	3.0
Max gas-jet velocity frac.	Ŷ	0.6	0.2	0.9



Pareto visualization – colored by vapor penetration merit





Liquid phase development

How correctly are we comparing measured and simulated liquid length?

- Exp datasets: Dashed line = penetration (Pickett et al., SAE2011-01-0686); Movie: Manin et al., COMODIA 2012
- How to distinguish between tiny droplets after catastrophic breakup and gas phase at the tip?



Mixture fraction distribution

- k-epsilon turbulence models suffer from jet over-dispersion (Pope, 1978)
- GA has tuned spray parameters to compensate for this → good mixing downstream and far from axis





Mixture preparation in the SNL light-duty engine

■ Acknowledgements! → Paul C. Miles, Dipankar Sahoo, Kan Zha, Stephen Busch



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Conclusions

- Advanced spray models implemented and validated in structured geometries
- Still room for improvement
 - Reduction of numeric constants in the gas-jet model ← Extension of the gas-jet theory
 - Reduction of numeric constants in the atomization model ← We range extension of the KH-RT model

Now need to apply the validated models in more comprehensive modeling of local flow field and thermal quantities → full engine geometry and full (multiple) cycle simulations

The model seems to be fast enough for it







Perini, F. and Reitz, R.D.,

"Improved atomization, collision and sub-grid scale momentum coupling models for transient vaporizing engine sprays"

Int J Multiphase Flow, submitted, 2015

see you at SAE...

PFL140 Fluid flow Measurement & Analysis

Wed Apr 22, 1:00 PM, Room 413 A

Perini, F., Zha, K., Busch, S., Miles, P.C., and Reitz, R.D.,

"Principal Component Analysis and Study of Port-Induced Swirl Structures in a Light-Duty Optical Diesel Engine"

SAE paper 2015-01-1696

